MEVISED 5-7-8

		HEATZED 2-1-B		
FMEA NO. W 11.14 CRITICALITY 2/18		SHUTTLE CCTV		
FAILURE MODE AND FAILURE EFFECT CAUSE ON END ITEM		RATIONALE FOR ACCEPTANCE		
CAUSE ON END ITEM No PTU control if PTU gets too cold. Worst Case:	PTU gets too cold. Worst Case: No PTU control of elbow camera which prevents	DESIGN FEATURES The Will RYS/PTU cable is a 15-inch long assembly, 16-wire assembly. The cable is terminated on each end with a 37-pin connector (Pl. KJGGE14N3SSN16). The vineo and swires are shielded #24 Twinax twisted-pair wires. The Will cable provides power and commands from the remote video switch (RYS) to the RMS elbow camera stack and returns video signals to the RYS. The cable design is taken from the successfully flown Apollo program. The design is cable—connector assembly in which the wire terminations are protected from excessive flexture at the joint between the wire and the connector terminal. The load concentration is moved away from the conductor connection and distributed axially alothe tength of the conductors encapsulated in a potted-taper profile. This technique also protects the assembly from dirt and entrapped molsture which could cause problem in space. The cable and its components meet the applicable requirements of NASA, Military and R specifications. These requirements include: General/Mechanical/Electrical Features Design and Construction Materials Terminal Solderability Environmental Qualification Marking and Serialization Traceability and Documentation		
		:		

REVISED 5-7-87

MEA NO. W 11.14 RITICALITY 2/1R		SHUTTLE CCTV CRETICAL LIENS LIST	UNII Cable DNG NO. 229395-502 LSSUED 10-14-86 SHEET 2 OF 5
ATLORE NODE AND CAUSE	FAILUHE EFFECT ON END ITEM	BATTOWALE FOR ACC	EPTANCE
of HTR RTN (PTU)	No PTU control if PTU gets too cold. Horst Case: No PTU control of elbow camera which prevents arm stowage.	Qualified by 1.) similarity to previous successfundification tests of CCTV LRUs. ACCEPTANCE TEST The cable acceptance test consists of an obmmeter connection is present and intact. Results are not operational tests OPERATIONAL TEST The following tests verify that CCTV components the PHS (A7A1) panel switch, through the RCU, but to the Camera/PTU command decoder are properability to produce video, the VSU's ability to display video. A similar test verifies the MDM Pre-Launch on Orbiter Test/In-Flight Test 1. Power CCTV System. 2. Select a monitor via the PHS panel, as desistance. 3. Send "Camera Power On" command from PHS panel, Select "External Sync" on monitor. If via stable raster), then this indicates that if from the RCU and that the camera is produce. 6. Send Pan, Tilt, Focus, Zoom, ALC, and Gamma monitor or direct observation) verify properation or direct observation verify properation of Camera Power Off" command via PHS panel. Send "Camera Power Off" command via PHS panel. Repeat Steps 3 through 9 except issue command proves that the CCTV equipment is operation.	er check to assure that each wire recorded on data sheets. are operable and that the commands from prough the sync lines to the Camera/PIU, The tests also verify the camera's route video and the munitor's ability to command path. tination and the camera under test as mel. dee on munitor is synchronized (i.e., me camera is receiving composite sync ing synchronized video. commands and visually (either via the er operation. under test as source. mel. ands via the MDM command path. This

CAUSE ON E of HTR RTR (PTU) No PTU core PTU gets to Norst Case No PTU core				
CAUSE ON E of HTR RTR (PTU) No PTU core PTU gets to No PTU core Camera whi		TLE CCIV L ITEMS LIST	UNIT CABTE UNG NO. 2293995-502 ISSUEO 10-14-86 SHEET 3 OF 5	
PTU gels t Norst Case No PTU cor camera whi	URE EFFECT END JTEM	RATIONALE FOR ACCEPTANCE		
	Procurement Control - Mi and suppliers which meet Plan Nork Statement (NS- incoming Inspection & St materials and parts. Re control numbers for futu Material Controlled Stor fabrication is required. (MRB) disposition. (PAI Assembly & Test - Prior by stock room personnel verified again by the op as-built-parts-list (ABP Instructions are given i 2280800 - Process Standa Standard in-line spilcin 2280876 - Process Standa Potting material and tes performed at the complet Preparation for Shipment packaged according to 22 All related documentatio is gathered and held in	the requirements set forth in 2593176]. orage - Incoming Quality inspensions are recorded by lot and me reference and traceability. Les and retained under specifie Mon-conforming materials are 1-307, PAI [QC-53]. to the start of assembly, all as the items are accumulated the crater who assembles the kit because who assembles the kit because the materials and and crimping flight connector of of standard intercornecting and marking of parts or assemble the procedure (TP-AT-2293287). It is not key operations. Then fabrication and test is 180746. Process Standard for Pa	theid for Material Review Board Items are verified to be correct to form a kit. The items are ty checking against the Applicable documents. These are contacts, 2280801 - Process wire using Raychem solder sleeves, ies with epoxy colors, 2280876. Quality and DCAS inspections are as complete, the cable assembly is takaging and Handling Guidelines. Parts List, ABPL, Test Data, etc.	

REV(SED 5-7-87

FMEA NO. W 11,14 CRITICALITY 2/JR		SHUTTLE CCTV CRITICAL LIENS LIST	UNET Cable DNG NO. 2293995-502 ISSUED 10-14-85 SHEET 4 OF 5
FATLURE HODE AND CAUSE DSS OF HTR ATN (PTU) Den	FAILURE EFFECT ON END ITEM No PTU control if PTU gets too cold. Worst Case: No PTU control of elbow camera which prevents arm stowage.	FATLURE HISTORY There have been no reported failures during RCA	
		•	
	. •		

REVISED 5-7-87

			, Melloco by dr
MEA NO. W 17.14 DETICALITY 2/1R		SINUTTLE CCTV CRITICAL ITEMS LIST	UNIT CABTE DWG NO. 2293995-502 ISSUED 10-14-86 SHEET 5 OF 5
ATLURE MODE AND CAUSE	FAILURE EFFECT ON END LIEM	DPERATIONAL EFFECIS Loss of ability to position the Elbow camera. Possible inability to stow the RMS if the elbow camera physically interferes with a payload. If RMS cannot be stowed the port payload bay door cannot be closed. Loss of trew and vehicle. CREW ACTIONS Perform EVA to reposition the elbow camera, use RMS motion to reposition the camera, or jettlson the RNS. CREW TRAINING Crew should be trained in contingency EVA and RMS operations procedures. MISSION CONSTRAINT On not manifest Elbow camera for any flight where the payload and the elbow camera can interfere with each other (for any pan or tilt angle). If the camera must be flown do not change the camera position until the interfering payload is deployed.	
OF KTR RTN (PTU)	No PTU control if pfN yets too cold. Norst Case: No PTU control of elbow camera which prevents arm stowage.		